



SERVICE BULLETIN 2006

DATE April 5, 1967

REVISIONS None

APPROVAL FAA

- I. SUBJECT: Corrosion Proofing of Steel Tubular Open Ended Structures
- II. PURPOSE: To Provide Corrosion Proofing of open end steel tubular structures which have not been corrosion proofed on the inside.
- III. AIRPLANE EFFECTIVITY: Aero Commander Model 200D, 301-383 inclusive.
- IV. COMPLIANCE: October 15, 1967, or next annual - whichever is sooner.
- V. REFERENCE DATA:
- 62C012 - Rudder Bellcrank
 - 63R001-7 - Cabin Sect. Tube
 - 64A015 - Main Gear Axle
 - 64A079 - Nose Landing Gear
 - 67R029-6 - Cowl Flap Tube
 - 67B083-5 - Cowl Flap Arm
 - 66B143-2 & 3 - Control Wheel Tube
 - 69C200 - Pitot Bracket
 - 66D037 - Rudder Bell Crank - Engine Mount
 - 66C152 - Bell Crank Flap Actuating
 - 64A039-2 - Gear Attach Pin-top
 - 61R002 - Center Wing Section -2-3-10-11 Tubes
 - 62A003 - Front Attach Fitting - Horizontal Stabilizer
- VI. PROCEDURES: The following steps are to be taken to insure adequate protection of the above listed open ended steel parts as referenced in Drawing #60B003.
- Step 1. The interiors of the above listed parts are to be coated or sprayed with a film of commercial grade, pure, boiled linseed oil - Sherwin Williams Company or equivalent is recommended.
- Step 2. Seal each open end of the steel tubular assemblies with 3M Body Caulking Compound #8547, Federal Specification #4-616-4.
- Step 3. Upon the completion of the above procedures make the proper entry in the A/C log book, fill out the compliance card, mail card to the manufacturer, and return the aircraft to service.